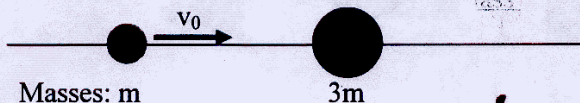


1. Assume this is a perfectly elastic head-on collision. Calculate the final velocity of the larger mass.

Solution Key



2. Consider a rocket in interstellar space. For $t < 0$ the rocket is at rest. At $t = 0$ the rocket is ignited. During the rocket burn, the mass exhaust rate $= \mu = -dm/dt$ is constant; and the relative exhaust speed is u . (Note that both μ and u are positive.) Determine the acceleration of the rocket a , as a function of time. Sketch a qualitatively correct graph of a versus t . Assume $m_{\text{final}} = 0.1 m_{\text{initial}}$.

(final velocities)

① Momentum is conserved $m v_0 = m v_1 + 3m v_2$ 1 point

Energy is conserved $\frac{1}{2} m v_0^2 = \frac{1}{2} m v_1^2 + \frac{1}{2} 3m v_2^2$ 1 point

Solve for v_2 :

$$v_1 = v_0 - 3v_2 \rightarrow v_0^2 = (v_0 - 3v_2)^2 + 3v_2^2$$

$$v_0^2 = v_0^2 - 6v_0 v_2 + 12v_2^2 + 3v_2^2$$

$$v_2 = \frac{1}{2} v_0 \quad \text{1 point}$$

(The section $v_2 = 0$ corresponds to no collision.)

3 points

② $\xrightarrow{\text{exhaust}} \rightarrow v$ Rocket equation $dP_{\text{total}} = 0$

$$dP_{\text{total}} = (m - \mu dt)(v + dv) + \mu dt (v - u) - mv = 0$$

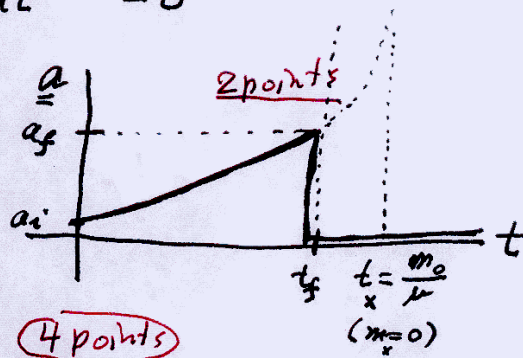
$$= m dv - \mu u dt = 0$$

$$a = \frac{dv}{dt} = \frac{\mu u}{m_0 - \mu t}$$

2 points

$$t_f = \frac{0.9 m_0}{\mu}$$

$$a_f = \frac{\mu u}{0.1 m_0}; \quad a_i = \frac{\mu u}{m_0}$$



HW Set F, Problem 1

Binary Star System

Consider the motion of M_2 -
Circular motion with
radius r_2 .

$$\frac{M_2 v^2}{r_2} = \frac{G M_1 M_2}{(r_1 + r_2)^2}$$

2 points

Note that $M_1 r_1 = M_2 r_2$

$$\begin{aligned} v^2 &= \frac{G M_1 M_2 r_2}{(r_1 + r_2)^2} \\ &= \frac{G M_1^2 R}{(M_1 + M_2) R^2} = \frac{G M_1^2}{(M_1 + M_2) R} \end{aligned}$$

Distances: $R = r_1 + r_2$
and $M_1 r_1 = M_2 r_2$

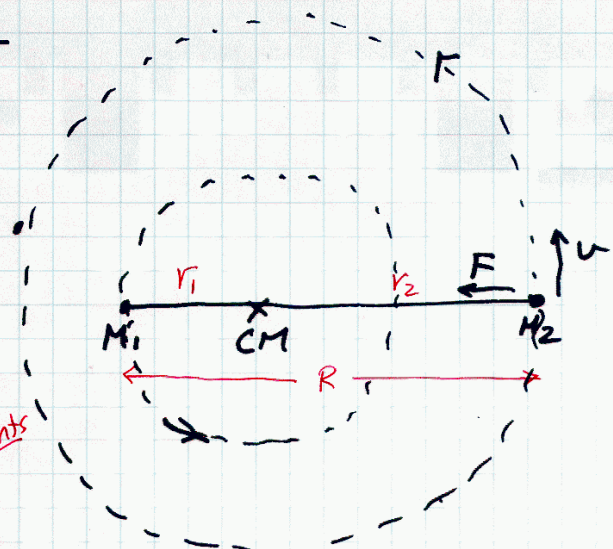
Therefore $r_1 = \frac{M_2 R}{M_1 + M_2}$ and $r_2 = \frac{M_1 R}{M_1 + M_2}$

The period of revolution of M_2 (which is equal
also to the period of revolution of M_1) is

$$T = \frac{2\pi r_2}{v} = \frac{2\pi M_1 R}{M_1 + M_2} \sqrt{\frac{(M_1 + M_2) R}{G M_1^2}}$$

$$\therefore T = \sqrt{\frac{4\pi^2 R^3}{G (M_1 + M_2)}} \quad \leftarrow 2 \text{ points}$$

4 points



HW Set F, Problem 10

Example 9.12 : Saturn rocket

(a) Velocity v versus time t for the first stage of the rocket, is given by Eq. (9.165)

$$v(t) = -gt + u \ln \frac{m_0}{m_0 - \mu t}$$

Use EXCEL, or some other computer graphics program, to make an accurate graph of $v(t)$. 2 points

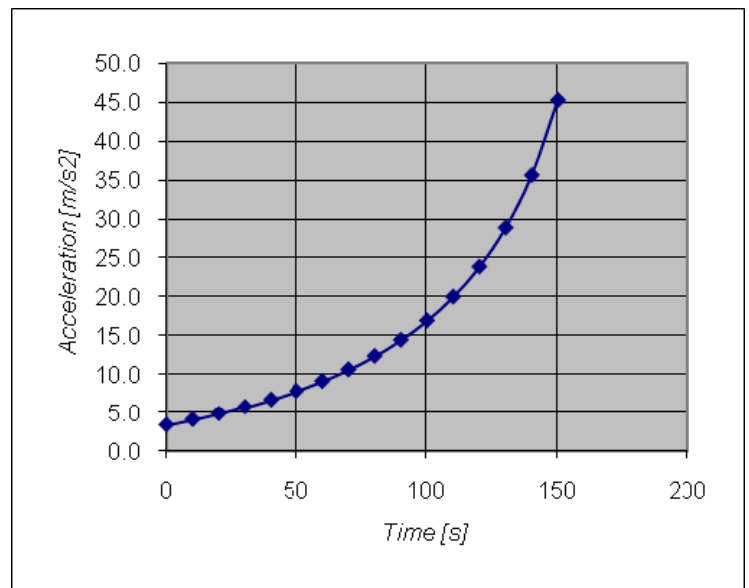
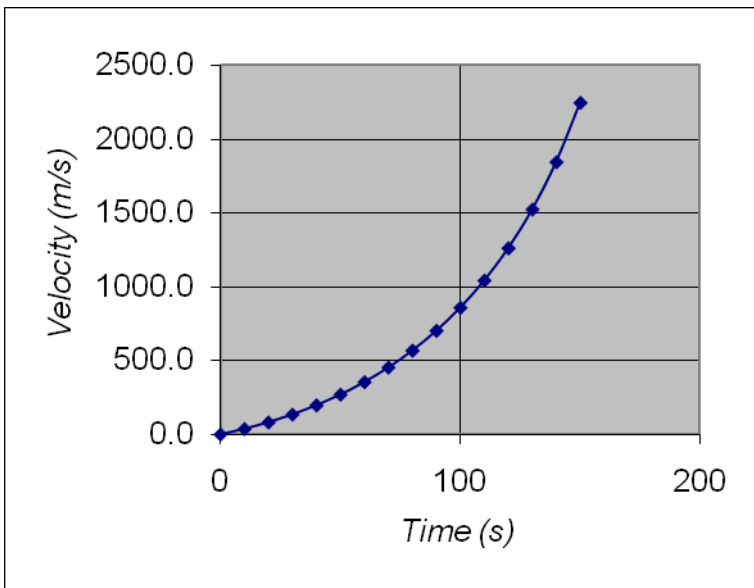
PARAMETERS	
g	$= 9.81 \text{ m/s}^2$
u	$= 2600 \text{ m/s}$
m_0	$= 2.8 \times 10^6 \text{ kg}$
μ	$= \overset{\text{exhaust}}{\text{mass rate}} = 1.42 \times 10^4 \text{ kg/s}$
t_B	$= \underset{\text{time}}{\text{burnout}} = 148 \text{ s}$

(b) The acceleration is

$$a(t) = \frac{dv}{dt} = -g + \frac{u\mu}{m_0 - \mu t}$$

Use EXCEL, or some other computer program to make an accurate graph of $a(t)$. 2 points

4 points



Linearly damped oscillator (Chap. 3)

$$m \ddot{x} = -\gamma \dot{x} - kx$$

$$\begin{aligned} \text{Try } x &= e^{\alpha t} \\ \dot{x} &= \alpha e^{\alpha t} \\ \ddot{x} &= \alpha^2 e^{\alpha t} \end{aligned}$$

$$m \alpha^2 = -\gamma \alpha - k$$

$$m \alpha^2 + \gamma \alpha + k = 0$$

$$\alpha = \frac{-\gamma \pm \sqrt{\gamma^2 - 4mk}}{2m}$$

Underdamped oscillations here $\gamma^2 < 4mk$.

$$\text{Then } \alpha = -\frac{\gamma}{2m} \pm i \sqrt{\frac{k}{m} - \left(\frac{\gamma}{2m}\right)^2}$$

$$\text{Define } \omega_0 = \sqrt{\frac{k}{m}} \quad \text{and} \quad \omega_d = \sqrt{\omega_0^2 - \left(\frac{\gamma}{2m}\right)^2}$$

$$\text{Then } \alpha = -\frac{\gamma}{2m} \pm i \omega_d$$

The solution with initial values $\begin{cases} x(0) = A \\ \dot{x}(0) = 0 \end{cases}$ is

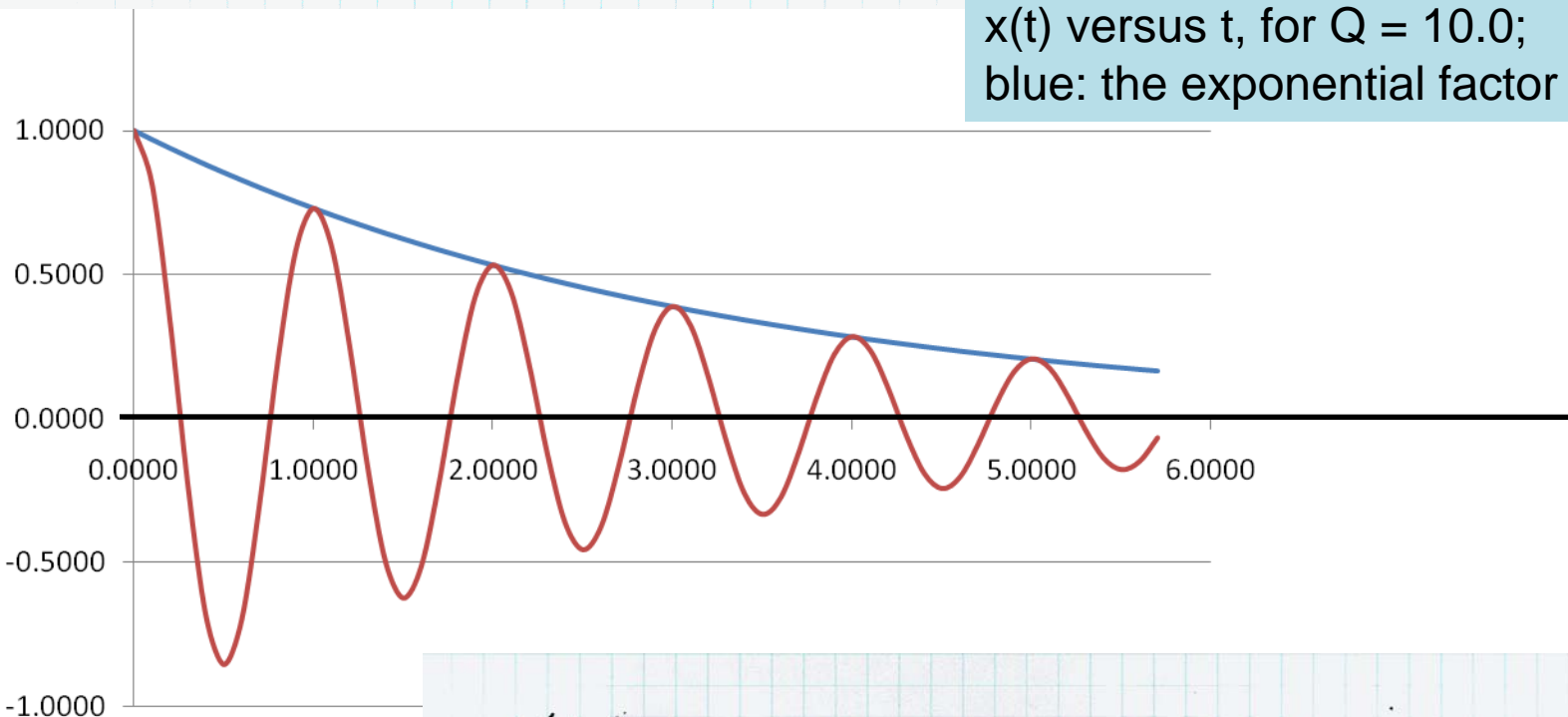
$$x(t) = A e^{-\gamma t / 2m} \left\{ \cos \omega_d t + \frac{\gamma}{2m \omega_d} \sin \omega_d t \right\}$$

$$\text{Euler } e^{i\theta} = \cos \theta + i \sin \theta$$

$$e^{\pm i \omega_d t} = \cos \omega_d t \pm i \sin \omega_d t$$

$$x(t) = A e^{-\gamma t/2m} \left\{ \cos \omega_d t + \frac{\gamma}{2m\omega_d} \sin \omega_d t \right\}$$

$x(t)$ versus t , for $Q = 10.0$;
blue: the exponential factor



Velocity After a bit of algebra ...

$$\dot{x}(t) = -A \omega_d \left[1 + \left(\frac{\gamma}{2m\omega_d} \right)^2 \right] e^{-\gamma t/2m} \sin \omega_d t$$

Note that $\dot{x}(t) = 0$ when $\omega_d t = \pi, 2\pi, 3\pi, 4\pi, 5\pi, \dots$

These are the times of maximum displacement from 0.

$x(t)$ is at a maximum positive displacement for

$$\omega_d t = 0, 2\pi, 4\pi, 6\pi, 8\pi, \dots$$

$$\omega_d t_n = 2\pi n \quad \text{where } n = 0, 1, 2, 3, \dots$$

$$x_n = A e^{-\gamma t_n/2m} \{ 1 + 0 \} = A e^{-\frac{2\pi\gamma}{2m\omega_d} n}$$

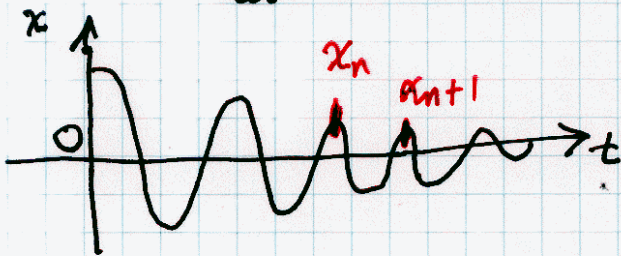
Energy and the underdamped oscillator

Define $E = \frac{1}{2} m \dot{x}^2 + \frac{1}{2} k x^2$.

If $\gamma = 0$ then E is constant.

If γ is "small" then E decreases slowly.

$$\frac{dE}{dt} = m \dot{x} \ddot{x} + k x \dot{x} = (m \ddot{x} + k x) \dot{x} = -\gamma \dot{x}^2$$



$$x_n = A e^{-\frac{\gamma}{m\omega_d} n}$$

$E_n = \frac{1}{2} k x_n^2 =$ the maximum of potential energy for positive displacements.

$$E_n = \frac{1}{2} k A^2 e^{-\frac{2\gamma}{m\omega_d} n} = \frac{1}{2} k A^2 e^{-2\pi n/Q}$$

Define $Q = \frac{m\omega_d}{\gamma}$

"Quality factor" is a dimensionless measure of the damping

Comments

- $Q = \frac{m\omega_d}{\gamma}$ where $\omega_d = \sqrt{\omega_0^2 - \left(\frac{\gamma}{2m}\right)^2}$

$$\text{So } Q = \sqrt{\left(\frac{m\omega_0}{\gamma}\right)^2 - \frac{1}{4}}$$

For weak damping, $Q \approx \frac{m\omega_0}{\gamma}$ and $Q \gg 1$.

- $\frac{E_n}{E_{n+1}} = \frac{e^{-2\pi n/Q}}{e^{-2\pi(n+1)/Q}} = e^{2\pi/Q}$ (independent of n)

$$Q = \frac{2\pi}{\ln(E_n/E_{n+1})}$$

- $E_{n+1} = E_n - |\Delta E|_n$ $|\Delta E|_n = \text{energy lost by friction in one cycle, from } n \text{ to } n+1.$
 $\frac{|\Delta E|_n}{E_n} = 1 - \frac{E_{n+1}}{E_n} = 1 - e^{-2\pi/Q}$

For $Q \gg 2\pi$ { i.e., $\gamma \ll \frac{m\omega_0}{2\pi}$; weak damping }

$$\frac{|\Delta E|_n}{E_n} \approx \frac{2\pi}{Q} \quad \text{or} \quad \frac{E_n}{|\Delta E|_n} \approx \frac{Q}{2\pi}$$

- Large Q means weak damping ("quality factor")