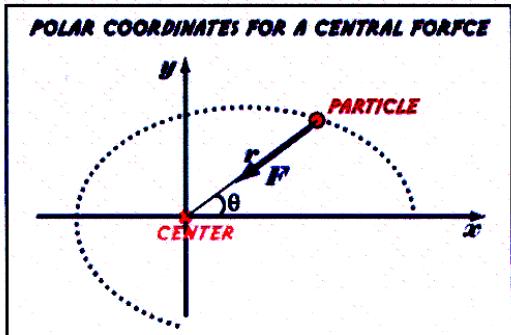


PHY 321 Quiz K August 4, 2009

A particle of mass m moves under the influence of a central force; $\mathbf{F} = -\nabla V$ where $V(r)$ is the potential energy. Angular momentum is conserved because the force is central. The constants of the motion are L and energy E .

We use polar coordinates $r(t)$ and $\theta(t)$ to describe the motion. *Derive an equation for $dr/d\theta$ as a function of r .* [The solution of this first-order differential equation determines the orbit.]



SOLUTION KEY

$$E = \frac{1}{2}m\dot{r}^2 + \frac{L^2}{2mr^2} + V(r) \quad | \text{ pt}$$

$$L = mr^2\dot{\theta} \quad | \text{ pt}$$

$$d\theta = \frac{L}{mr^2} dt$$

$$dr = \pm \sqrt{\frac{2}{m}(E - V(r))} \quad \text{and also } V_{\text{eff}}(r) = V(r) + \frac{L^2}{2mr^2} \quad) \text{ method } | \text{ pt}$$

$$\frac{dr}{d\theta} = \pm \frac{mr^2}{L} \sqrt{\frac{2}{m}(E - V_{\text{eff}}(r))}$$

~~Method~~

~~Method~~
answer | pt

5 pts

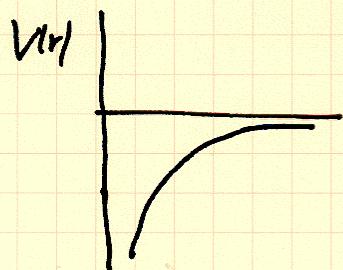
$$\frac{dr}{d\theta} = \pm \frac{mr^2}{L} \sqrt{\frac{2}{m}(E - V_{\text{eff}}(r))}$$

when $V_{\text{eff}}(r) = V(r) + \frac{L^2}{2mr^2}$

Since L is constant

Problem K904

(Handed in with H.W. Set K)

Let $V(r) = -\frac{A}{r^n}$ where $A > 0$.

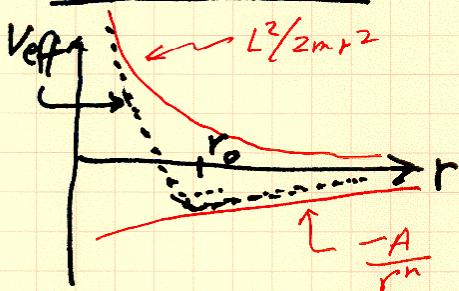
The force is directed toward $r=0$.

For what values of n will the circular orbits be stable?

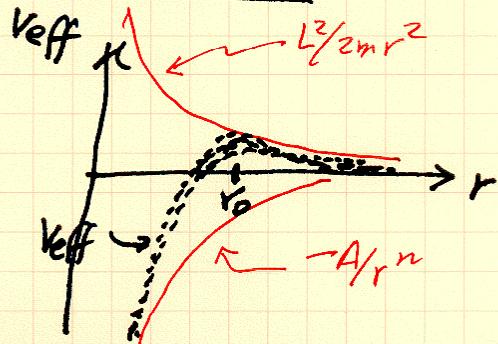
Orbit Equations

$$L = mr^2 \dot{\phi} = \text{const.}$$

$$\begin{aligned} E &= \frac{1}{2}m\dot{r}^2 + \frac{1}{2}m(r^2\dot{\phi}^2) + V(r) = \text{const.} \\ &= \frac{1}{2}m\dot{r}^2 + V(r) + \underbrace{\frac{L^2}{2mr^2}}_{\text{constant}} \\ &= \frac{1}{2}m\dot{r}^2 + V_{\text{eff}}(r) \end{aligned}$$

Case $n < 2$ 

The circular orbit ($r=r_0$) is stable.

Case $n > 2$ 

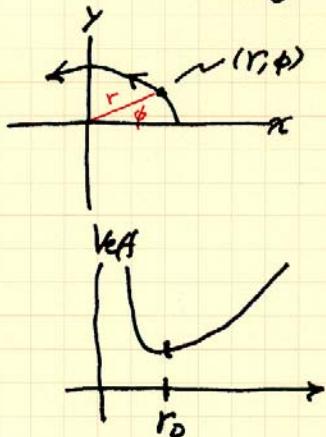
The circular orbit ($r=r_0$) is unstable.

The circular orbits are stable if $n < 2$.

Problem K906

(Handed in with HW Set K)

A particle of mass m moves under the influence of potential energy $V(r) = \frac{1}{2}kr^2$.



$$L = mr^2\dot{\phi} = \text{constant}$$

$$E = \frac{1}{2}m\dot{r}^2 + V_{\text{eff}}(r) = \text{const.}$$

$$V_{\text{eff}}(r) = V(r) + \frac{L^2}{2mr^2}$$

$$V_{\text{eff}}(r) = \frac{1}{2}kr^2 + \frac{L^2}{2mr^2}$$

The circular orbit has $r(t) = r_0$ where $\frac{dV_{\text{eff}}}{dr} = 0$;

$$\text{i.e., } kr_0 - \frac{L^2}{mr_0^3} = 0$$

$$r_0 = \left(\frac{L^2}{mk}\right)^{1/4} \text{ and } E = V_{\text{eff}}(r_0) = \left(\frac{kL^2}{m}\right)^{1/2}.$$

Now consider small oscillations around the circular orbit.

Write $r(t) = r_0 + \epsilon(t)$

$$V_{\text{eff}}(r) \approx V_{\text{eff}}(r_0) + \frac{1}{2}\epsilon^2 \left(\frac{d^2V_{\text{eff}}}{dr^2}\right) = C + \frac{1}{2}K\epsilon^2$$

$$\text{Here } K = V''_{\text{eff}} = k + \frac{3L^2}{mr_0^4} = 4k.$$

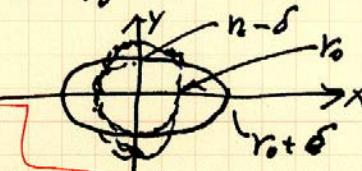
The radial frequency is $\omega_r = \sqrt{\frac{K}{m}} = 2\sqrt{\frac{k}{m}}$ //HARMONIC RADIAL OSCILLATIONS//

The orbital frequency is $\omega_\phi = \dot{\phi} = \frac{L}{mr_0^2} = \frac{L}{m} \left(\frac{mk}{L^2}\right)^{1/2} = \sqrt{\frac{k}{m}}$

Result $\omega_r = 2\omega_\phi$

What does that say about the orbit?

The orbit is approximately an ellipse



IN FACT, the orbits are

exactly ellipses. Do you see why?

$m\ddot{x} = -kx$ and $m\ddot{y} = -ky$ in Cartesian coordinates.

Quiz L (8/11/09)

Keplerian Orbits

To determine the orbits, we use the constants of the motion. There are two physical parameters: m = the mass of the planet; and K = the force parameter = GMm .

$$L = mr^2 \dot{\theta}$$

$$E = \frac{1}{2}mr^2 + \frac{L^2}{2mr^2} - \frac{K}{r}$$

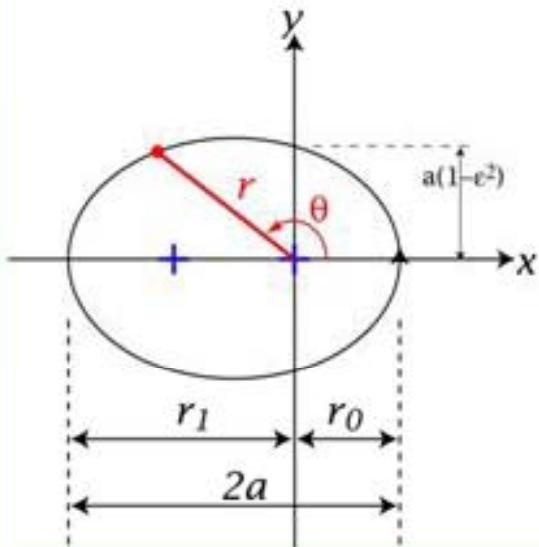
Now derive a differential equation for the distance r versus angle θ . Then it can be shown that the solution for r as a function of θ is

$$\frac{1}{r} = \frac{mk}{L^2} + A \cos \theta \quad \text{where } A^2 = \frac{2mE}{L^2} + \left(\frac{mk}{L^2}\right)^2$$

$$\frac{dr}{d\theta} = \frac{dr/dt}{d\theta/dt} = \frac{mr^2}{L} \sqrt{\frac{2}{m} \left(E - \frac{L^2}{2mr^2} + \frac{K}{r}\right)}$$

This is the equation for an ellipse, with one focal point at $r = 0$.

Ellipse Geometry



r_0 = perihelion distance ($= r$ at $\theta = 0$)

r_1 = aphelion distance ($= r$ at $\theta = \pi$)

a = semi-major axis $= (r_1 + r_0)/2$

e = eccentricity $= (r_1 - r_0)/(r_1 + r_0)$

$r_0 = a(1 - e)$ and $r_1 = a(1 + e)$

The equation for the ellipse is

$$r(\theta) = \frac{a(1 - e^2)}{1 + e \cos \theta}$$

(A) Express L as a function of (a, e) {and the parameters m, K }

(B) Express E as a function of (a, e) {and the parameters m, K }.