## PHY321 Homework Set 5

1. [5 pts] Determine the differential cross section $\sigma(\theta) \equiv \mathrm{d} \sigma / \mathrm{d} \Omega$ and total cross section $\sigma_{t}$ for elastic scattering of a point particle from a strong repulsive potential sphere of radius $R$ :

$$
U(r)= \begin{cases}0, & r<R, \\ U_{0}, & r>R,\end{cases}
$$

where $U_{0} \rightarrow \infty$. For a particle scattered from a strong potential, the law of reflection is valid, see the figure. Hint: From geometry work out the relation between the scattering angle $\theta$ and the impact parameter $b$.

2. [10 pts] Start with the Rutherford scattering cross section

$$
\frac{\mathrm{d} \sigma}{\mathrm{~d} \Omega^{\prime}}=\frac{1}{16}\left(\frac{k q_{1} q_{2}}{T^{\prime}}\right)^{2} \frac{1}{\sin ^{4}\left(\theta^{\prime} / 2\right)},
$$

where $k=1 / 4 \pi \epsilon_{0}$ and $T^{\prime}$ is kinetic energy within the center of mass of the charges $q_{1}$ and $q_{2}$.
(a) By integrating the above differential cross section, obtain the net cross section for scattering to angles $\theta^{\prime}>\theta_{0}$ :

$$
\sigma_{\theta^{\prime}>\theta_{0}}=\int_{\theta^{\prime}>\theta_{0}} \frac{\mathrm{~d} \sigma}{\mathrm{~d} \Omega^{\prime}} \mathrm{d} \Omega^{\prime} .
$$

Hint: When a solid-angle integration extends over the full range of azimuthal angle, and there is no dependence on the latter angle in the subintegral function, then the suitable element of solid-angle integration is $\mathrm{d} \Omega^{\prime} \equiv 2 \pi \sin \theta^{\prime} \mathrm{d} \theta^{\prime}$. In the particular case, it may be further useful to change the variable of integration from $\theta^{\prime}$ to $\sin \left(\theta^{\prime} / 2\right)$.
(b) Find the net cross sections for Rutherford scattering of 7.7 MeV alpha particles from gold nuclei, through angles I: $\theta^{\prime}$ greater than $170^{\circ}$, II: greater than $90^{\circ}$, and III: greater than $10^{\circ}$. Express your cross sections both in $\mathrm{m}^{2}$ and in barns; $1 \mathrm{~b}=10^{-28} \mathrm{~m}^{2}=100 \mathrm{fm}^{2}$ represents approximately the size of uranium nucleus. Data: $m_{\alpha}=4 \mathrm{u}, Z_{\alpha}=2, m_{\mathrm{Au}}=197 \mathrm{u}$ and $Z_{\mathrm{Au}}=79 ; e^{2} / 4 \pi \epsilon_{0}=1.4400 \mathrm{MeV} \mathrm{fm}$. For $m_{\alpha} \ll m_{\mathrm{Au}}$, you can in practice ignore the difference between laboratory and center-of-mass reference frames.
3. [5 pts] A rocket of mass $m_{0}$ starts its engine in interstellar space. Assuming constant speed $u$ of the exhaust gas relative to the rocket, at what fraction of the original mass is the rocket going to achieve maximal momentum?
4. [10 pts] Consider the problem of a rocket ascending vertically against gravity. The rocket starts from rest and its initial mass is $m_{0}$. The rocket's fuel burns at a constant rate $\alpha$ and exhaust gas leaves the rocket at a constant speed $u$ relative to the rocket. A convenient characteristic of a rocket, that is commonly used in place of $\alpha$, is the initial trust-to-weight ratio $\tau_{0}=\alpha u / m_{0} g$.
(a) From the expression for the velocity of the rocket as a function of the remaining mass $m$,

$$
v=-\left(m_{0}-m\right) \frac{g}{\alpha}+u \ln \left(\frac{m_{0}}{m}\right)
$$

eliminate $\alpha$ and write the velocity in terms of $u$, the mass ratio $m_{0} / m$ and $\tau_{0}$.
(b) Demonstrate that, for the lift-off to occur, the rocket must be light enough so that $\tau_{0}>1$.
(c) Integrate the velocity with respect to time to obtain elevation $h$ of the rocket as a function of the remaining mass $m$. Note that the integration with respect to time can be easily converted to integration with respect to mass exploiting the linear relation between the two variables. Note further that $\int \mathrm{d} x \ln x=x \ln x-x$. Again eliminate $\alpha$ from your result and represent $h$ in terms of $\tau_{0}, u, g$ and the mass ratio $m_{0} / m$.
(d) Consider the case of Ariane 5 rocket with initial mass of $m_{0}=7.77 \times 10^{5} \mathrm{~kg}$. During the initial stage-0 of the flight, boosters are used that provide a thrust of $\alpha u=1.29 \times 10^{7} \mathrm{~N}$ and employ solid fuel with exhaust velocity of $u=3010 \mathrm{~m} / \mathrm{s}$. At the end of stage- 0 , the rocket mass drops to $m=2.23 \times 10^{5} \mathrm{~kg}$. Find $\tau_{0}$ and mass ratio $m_{0} / m$. Use those to determine the expected velocity and elevation of the rocket at the end of stage- 0 .
5. [10 pts] A uniform rope of mass $M$ and length $L$ is hung off a small peg. The rope can slide without friction over the peg.
(a) With $x$ denoting the length of rope hanging to the right of the peg, obtain an equation of motion for $x(t)$.
(b) Solve the equation of motion, if an $x_{0}$-stretch of the rope hangs to the right of the peg at $t=0$ and the rope is then at rest. How does the fate of the rope depend on whether $x_{0}>L / 2$ or $x_{0}<L / 2$ ?
(c) Find the time $t_{f}$ that it takes for the rope to fall off the peg. How does this time depend on $M$ ?
6. [5 pts] A massless spring hangs down from a support, with its lower end at $y=0$, where the $y$-axis is vertical and points down. When a small unknown mass is attached to the spring, the lower end of the spring moves down to a position $y_{0}$ for the mass being in equilibrium.
(a) Demonstrate that when the mass is pulled down to a position $y=$ $y_{0}+A$ and released from rest, it will execute a simple harmonic motion around $y_{0}$.
(b) Express the period of oscillations of the mass in terms of $y_{0}$ and $g$.


